Post-shock temperatures in minerals

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Received 1979 February 5; in original form 1978 September 12

Summary. An experimental technique has been developed for the measurement of post-shock temperatures in a wide variety of materials, including those of geophysical interest such as silicates. The technique uses an infrared radiation detector to determine the brightness temperature of samples shocked to pressures in the range 5 to ~30 GPa; in these experiments measurements have been made in two wavelength ranges (4.5 to 5.75 μm; 7 to 14 μm). Reproducible results, with the temperatures in the two wavelength bands generally in excellent agreement, have been obtained for aluminium-2024 (10.5 to 33 GPa; 125 to 260°C), stainless steel-304 (11.5 to 50 GPa; 80 to 350°C), crystalline quartz (5.0 to 21.5 GPa; 80 to 250°C), forsterite (7.5 to 28.0 GPa; -30 to 160°C) and Bamble bronzite (6.0 to 26.0 GPa; ~30 to 225°C).

These results are generally much higher at low pressures (where they may even be in excess of the calculated shock temperatures) than the values calculated assuming a hydrodynamic rheology and isentropic release parallel to the Hugoniot but tend towards them at higher pressures. In aluminium-2024, the theoretical post-shock temperatures, assuming a fluid-like rheology, are 35 to 218°C, for the pressure range 10.5 to 33 GPa. However, the results are in considerably better agreement with values calculated assuming elasto-plastic behaviour (80 to 270°C) which probably also causes the high measured temperatures for stainless steel. In forsterite the measured values ranged from 65°C at 9.6 GPa (there was no detectable rise at 7.5 GPa) to 156°C at 28.0 GPa, whereas the 'hydrodynamic values' were 30 to 120°C. Values obtained for quartz were in excellent agreement with those calculated by Mashimo et al. using release adiabat data. It is concluded that release adiabat data should be used, wherever available, for calculations of residual temperature, and that adequate descriptions of the shock and release processes in minerals need to be more complex than generally assumed.

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Introduction

Shock wave equation of state data have long been used in the interpretation of impact metamorphism (e.g. Stöffler 1971, 1972) and of density depth profiles of the Earth obtained from seismic data (e.g. Al'tschuler 1965). However, one of the limitations to these uses of Hugoniot data is the uncertainty in the temperatures reached both during the passage of the shock wave through the material and after unloading.

Shock temperatures have previously been calculated using Hugoniot data and the Mie–Grüneisen theory (e.g. Walsh & Christian 1955; Wackerle 1962; Ahrens, Anderson & Ringwood 1969; McQueen et al. 1970) which should yield reliable results for metals provided the rheology corresponds to fluid-like behaviour. However, the Mie–Grüneisen theory, when the material is treated as a single phase, is inadequate over much of the pressure range for which shock wave data exist for silicates, since these all undergo major phase changes which may involve substantial changes in thermodynamic properties and internal energy. Calculated Hugoniot temperatures can then be used as a starting point to obtain post-shock temperatures assuming isentropic release. Unfortunately, there are few cases where there are sufficient data either to determine experimentally or calculate theoretically isentropic release paths; moreover the assumption of isentropic unloading from shock states has never been explicitly verified experimentally. Consequently, even if the Hugoniot temperatures were calculated correctly, large uncertainties could still exist in residual temperatures.

Calculations of post-shock temperatures in silicates, assuming release along isentropes lying above the Hugoniot (Fig. 1), with the release volume greater than the initial volume, lead to values that appear too low to account for some of the effects seen in recovery experiments (e.g. Gibbons 1974). Where release adiabat data exist for silicates, they have been used in the calculation of post-shock temperatures (e.g. Gibbons & Ahrens 1971; Ahrens & O'Keefe 1972). This always leads to much higher, and possibly more credible, release temperatures (as is shown by the comparison in Table 1) largely because the release paths lie below the Hugoniot.

Figure 1. Schematic diagram of Hugoniot and possible release adiabats. The solid release curves lie above the Hugoniot, and the dashed ones below it.
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Table 1. Calculated post-shock temperatures in fused quartz and oligoclase, using the Mie–Grüneisen theory (a) or release adiabat data (b).

<table>
<thead>
<tr>
<th>Shock pressure (GPa)</th>
<th>Post-shock temperatures, °C</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>Fused quartz</td>
</tr>
<tr>
<td>10.0</td>
<td>0</td>
</tr>
<tr>
<td>15.0</td>
<td>0</td>
</tr>
<tr>
<td>18.0</td>
<td></td>
</tr>
<tr>
<td>25.0</td>
<td>0</td>
</tr>
<tr>
<td>27.2</td>
<td>470</td>
</tr>
<tr>
<td>30.0</td>
<td>1860</td>
</tr>
<tr>
<td>41.7</td>
<td>3310</td>
</tr>
<tr>
<td>50.0</td>
<td></td>
</tr>
</tbody>
</table>

¹ Wackerle (1962).
² Gibbons & Ahrens (1971).
³ Ahrens et al. (1969b).
⁴ Ahrens & O'Keefe (1972).

Early experiments to measure post-shock temperatures were limited to metals; using a photo-multiplier tube Taylor (1963) found values for copper shocked to pressures in the range 90 to 170 GPa that were in good agreement with those calculated by McQueen & Marsh (1960) using Mie–Grüneisen theory and fluid rheology, although later work by King, Cotgrove & Slate (1968) and Von Holle & Trimble (1976) using infrared radiation detectors revealed considerable discrepancies between observed and calculated temperatures for copper shocked to pressures below 80 GPa. However, no attempts were made to extend this type of experiment to non-metals, largely because the supposedly lower temperatures, coupled with the low sensitivity of available detectors, meant that experiments such as those of King et al. were not feasible. Recent improvements in detector technology have now made it possible to design a system capable of measuring post-shock temperatures in silicates; the availability of such data should help resolve the current uncertainties.

Experimental technique

Initial experiments were designed with the aim of investigating residual temperatures in silicates of geophysical interest shocked to pressures up to 30 GPa. The materials chosen were natural crystal quartz (cut perpendicular to the c axis), Bamble bronzite (from Bamble, Norway), which consists of large natural single crystals closely described by the formula (Mg₀.₈₆Fe₀.₁₄)SiO₃ which have a porosity of 1 to 3 per cent, and synthetic crystal forsterite (cut perpendicular to the c axis). For completeness, the materials used as driver plates in the experiments, namely aluminium-2024 and stainless steel-304, were also studied.

The method developed involves the monitoring of radiation from the back (free) surface of a shocked sample with an infrared radiation detector whose output may then be used to determine the brightness temperature of the sample. Since the residual temperatures for the pressure range to be investigated were expected to be of the order of 400 K, and the Planck distribution law gives a maximum in spectral radiance between 3 and 9 µm for blackbodies radiating at temperatures from 1000 to 300 K (Touloukian & DeWitt 1972) infrared detectors were a logical choice for this study. In addition, since silicates behave as fairly good blackbodies in the infrared beyond ~5 µm, with the exception of a minimum in
emissivity at $\sim 9 \, \mu m$ (Lyon 1965), the radiative output of the sample is also maximized. A schematic plan of the experimental configuration is shown in Fig. 2.

(a) PRODUCTION OF THE SHOCKED STATE

In these experiments, the shocked state was produced in the sample by the impact of a gun-launched flyer plate. The technique is described in detail by Gibbons (1974) and Raikes (1978) and has been used by a number of authors (e.g. Ahrens, Lower & Lagus 1971; Ahrens & Gaffney 1971; King & Ahrens 1976). The velocity of the projectile is determined from the interval between obscurations (by the front of the projectile) of the two laser beams (Fig. 2); typical uncertainties in velocity are $\pm 0.05 \, km/s$. The shock pressure in the sample is then calculated using the impedance match method (e.g. Duvall & Fowles 1963) and the known Hugoniots of flyer plate, driver plate and sample materials. Estimates of the uncertainty in pressure determination, based on the accuracy of velocity measurement and scatter of the Hugoniot points, are $0.5$ to $1.0 \, GPa$ in the silicates.

After passing through the mylar window sealing the end of the barrel, the projectile makes contact across a shorting target made from thin copper foil and having a negligible effect on the projectile velocity, and finally impacts the target assembly (Fig. 2). In order to achieve nearly one-dimensional planar flow upon impact of the projectile with the target, the target is carefully aligned using the normal reflection from the target surface of a laser beam shining down the centre of the barrel. The target assembly consists of a driver plate (1.5 mm aluminium-2024 or stainless steel-304), followed by a 3 mm thick silicate sample, 15 mm in diameter; in the case of shots on metals the driver plate, now 3 mm thick, is the sample.

(b) TEMPERATURE MEASUREMENT

The brightness temperature of the back (free) surface of the sample is determined from the output from an infrared detector. This detector is mounted above the impact chamber (Fig. 2), and monitors the back face of the sample via a mirror and optical system. The latter
ensures that only the centre (~0.8 cm dia) of the sample is viewed, reducing the contribution from radial release waves (also somewhat lessened by using a circular sample) and increases the efficiency of the detection system. The detector is connected via an amplifier to two oscilloscopes. One is triggered by the passage of the projectile past the first laser beam of the timing system, and records the detector output at a rate of 50 µs/div. This provides a back-up record in case of failure of the higher-resolution recording, and a means of checking that no temperature signals are generated prior to the passage of the shock wave through the sample assembly. (Note that it also provides another means of determining the projectile velocity.) The second oscilloscope is triggered by the contact of the flyer plate with the shorting target, which is approximately 15 mm in front of the driver plate, just prior to impact; this writes at 5 µs/div, and it is the primary record that is used in temperature determination. The interpretation of this record is based on the assumption that the oscilloscope sweep is triggered at the moment of contact of the flyer plate with the shorting target; this was checked by analysing the timing of signals seen on the back-up record (50 µs/div) and by using a delayed trigger on the oscilloscope, and appears valid to within ±0.5 µs, which is close to the rise time of the systems. A typical record (such as that for bronzite at 15.5 GPa, Fig. 8(c)) shows a sharp rise in signal level as the shock wave arrives at the free surface of the sample followed by a level portion corresponding to the residual temperature, and then a subsequent rise due to air shocks generated at the end of the sample chamber and the destruction of the mirror. At high pressures, especially for metals which have a second peak in the record (Fig. 7), the level portion may not be clearly visible; in this case the post-shock temperature is determined from the signal 0.5 µs (InSb) or 0.75 µs (HgCdTe) after the free surface arrival. (These values are the rise times of the detection systems.) Problems in resolving the residual temperature were also encountered for quartz where an initial flash (Fig. 8(a)) made identification of the level portion uncertain at the highest pressures.

In InSb and HgCdTe detectors were operated at 77 K. They are enclosed in dewars cooled by liquid nitrogen; both were obtained from the Santa Barbara Research Centre (Goleta, California) and typical response curves, obtained from SBRC, are shown in Fig. 3. Additional operational details are listed in Appendix A. A filter was used to limit the bandwidth of the InSb detector to 4.5 to 5.75 µm in order to minimize the possibility of radiation from the metal driver plate, or metal—sample interface, being transmitted through the sample and causing errors in the temperature determination. (As can be seen from the

Figure 3. (a) Response curve for InSb detector material. (b) Response curve for HgCdTe detector material. $D^*(\lambda, \nu)$ is the area independent sensitivity at frequency $\nu$ and wavelength $\lambda$. Shaded areas indicate typical variations in detector sensitivity. (Details from the Santa Barbara Research Centre Catalogue.)
transmission scans of Fig. 4, which were obtained using a Perkin-Elmer Model 180 infrared spectro-photometer in a manner analogous to that described in Burns (1970), this is unlikely to be a problem for quartz or bronzite but the forsterite does transmit significantly up to 5 μm.)

The InSb detector was used with a variable gain amplifier (Raikes 1978); the system rise time, which is essentially limited to ~0.2 μs by the chip itself, was ≤0.5 μs depending on the gain setting used. Although the HgCdTe is a faster material (≤0.1 μs), the rise time of the detector-amplifier system was ≤0.8 μs because an electronic filter with a high-frequency cut-off of 5 MHz had to be used to reduce the large amount of high-frequency noise which would otherwise have made accurate measurement of temperature impossible.

In order to convert the voltage record into a brightness temperature, the detector must be calibrated. This is best performed by heating the sample in situ to a known temperature, and recording the corresponding voltage output of the detector-amplifier system. This is easily done for metals, but would be very hard for the non-metals studied since they are extremely brittle and hard to heat in the experimental configuration without cracking. Instead, initial estimates of temperature were based on the assumption that the silicates behaved as black bodies using a calibration curve obtained for a "black" body, graphite. For quartz in range 5 to 8 μm this is a reasonable approximation since the emissivity is greater than 0.9 (Touloukian & DeWitt 1972), and is probably justified for the Bamble bronzite in the InSb range, since it does not transmit. However, the forsterite has a 20 per cent transmittance at 4.5 μm (dropping rapidly to less than 5 per cent at 5 μm), and all the silicates studied have a large drop in emissivity at around 9 μm. This is clearly visible in the comparison of blackbody and quartz emittance spectra at temperatures from 250 to 500 K presented by Lyon (1965). Corrections for sample emissivity were determined using available data which are, unfortunately, largely for powdered minerals or rocks. The values of the emissivity for quartz and dunite (primarily forsterite) given by Buettner & Kern (1965), Lyon (1965) and Touloukian & DeWitt (1972) were used to calculate correction factors as follows.

The detector output $S$ for an operating wavelength range from $\lambda_1$ to $\lambda_2$ μm may be expressed as

$$S = \int_{\lambda_1}^{\lambda_2} E(\lambda, T)D(\lambda)P(\lambda, T) d\lambda$$

(1)

where $E(\lambda, T) = E(\lambda) =$ emissivity (assumed independent of $T$), $D(\lambda) =$ detector response, $P(\lambda, T) =$ Planck's function, $\lambda =$ wavelength, $T =$ absolute temperature.
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Table 2. Estimated signal correction factors. Correction factor = silicate signal/blackbody signal (values to nearest 0.05).

<table>
<thead>
<tr>
<th></th>
<th>$T = 400$ K</th>
<th>$T = 600$ K</th>
</tr>
</thead>
<tbody>
<tr>
<td>InSb</td>
<td>HgCdTe</td>
<td>InSb</td>
</tr>
<tr>
<td>Quartz (polished)</td>
<td>0.90</td>
<td>0.80</td>
</tr>
<tr>
<td>Quartz (rough)</td>
<td>0.90</td>
<td>0.90</td>
</tr>
<tr>
<td>SiO$_2$</td>
<td>0.90</td>
<td>0.90</td>
</tr>
<tr>
<td>Dunite (polished)</td>
<td>0.85</td>
<td>0.90</td>
</tr>
<tr>
<td>(Bronzite)</td>
<td>0.9</td>
<td>0.85</td>
</tr>
</tbody>
</table>

This may be integrated numerically and used to derive the ratio of the signal obtained for a silicate at temperature $T$ to that for a blackbody at the same temperature. The correction factors are listed in Table 2; the values for bronzite were estimated from those for quartz and dunite based on a comparison of absorption spectra of the three minerals.

Since the post-shock temperatures are measured after the interaction of the shock-wave with the free surface, and that interaction will cause roughening of the surface, a correction factor derived for a rough surface might be more appropriate; however, the values listed in Table 2 for polished surfaces will be used since these should yield an upper bound on the temperature. Two additional factors should be taken into account: one is that the absorption peak may shift during shock compression (e.g. Goto, Rossman & Ahrens 1979), and broadening of the absorption bands for SiO$_2$ has been observed in samples recovered after shock compression to pressures up to 52 GPa (Mashimo et al. 1979; Stöffler 1974). The second is the possibility of triboluminescence, or some other form of non-equilibrium radiation such as might be associated with a phase change; in these cases the emissivity may even exceed unity. Because of these uncertainties, the blackbody temperatures may well be more reasonable estimates of the residual values and will be used later in a comparison of observed and calculated temperatures; the corrected values probably represent upper bounds to the post-shock temperatures.

![Detector Calibration](http://gji.oxfordjournals.org/)

Figure 5. Calibration curves for the InSb detector, operating in the wavelength range 4.5 to 5.75 μm. The open and solid circles in the aluminium curve are the results of two different calibration runs.
Typical calibration curves are plotted in Figs 5 and 6, and Table 3 gives the power law fits to the curves. Both detectors gave extremely reproducible calibration curves, as is demonstrated by the two sets of points for aluminium in the case of InSb (these were obtained several weeks apart with a number of shots fired in between).

Sources of error

Errors in temperature measurement can basically arise from two causes – those related to the sample, and those originating from outside sources. Prime among the latter is contamination of the signal by radiation from air shocks which can be of extremely high temperature. Owing to the geometry of the apparatus these should not be important prior to the arrival of the shock wave at the free surface of the sample and, indeed, no earlier signal rises were detected. The only air shock likely to affect the post-shock temperature
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measurement would be one generated at the back of the sample itself, which should be eliminated by the sample vacuum chamber that is pumped down to ≤ 5 m torr (≤ 0.7 Pa). To reduce further possible radiation from residual gases within this chamber heated by compression due to the shock wave, the chamber was flushed out with helium prior to each shot. Radiation from later air shocks, such as that generated at the end of the sample chamber as the window breaks due to the shock wave travelling down the chamber walls and impact of ejecta from the samples, is clearly visible on each record, and ultimately causes the detection system to saturate. If the samples were transparent, then radiation from the metal driver plate could add to the signal, but the rise should then precede the free surface arrival of the shock wave, and this does not appear to be the case. Anyway, the emissivity of the metal is substantially lower than that of the sample, so this effect should be relatively small, and both detectors were chosen to operate at wavelengths where the samples are nearly opaque. (Forsterite does have a 20 per cent transmittance at 4.5 μm dropping rapidly to less than 5 per cent at 5 μm, and so may show minor effects due to transmitted radiation for the InSb detector.) Both the driver plate and the sample surface in contact with it were polished prior to mounting of the sample in order to minimize the 'porous' surface interactions that could give rise to considerable heating (see, e.g. Urtiew & Grover 1974); this also reduces the likelihood of air being trapped in this interface, but this should be removed by the evacuation of the sample chamber. In order for the signal to be contaminated by radiation from the metal or the driver—sample interface, large changes must take place in the transmissivity of the sample under shock conditions; although changes have been reported in sapphire (Urtiew 1974), these were decreases and at much higher pressures. It seems that this is not a likely source of error. Vibration of the detector-amplifier system could conceivably affect the output. However, the detection system was physically clamped in isolation from the gun, and vibrational effects were not apparent except for some HgCdTe shots where a negative signal of short duration (~ 10 μs), obviously non-thermal in origin, was observed prior to impact with the shorting target, but the output returned to the zero level before the free surface arrival.

Two main sources of errors associated with the sample behaviour under shock are changes in emissivity and the effects of non-uniform heating. The former may be investigated by comparing the brightness temperatures obtained at different wavelengths, since the change would probably not be constant as a function of wavelength. Changes in emissivity may be related to phase changes, changes in surface properties and triboluminescence. In a recent work on residual temperatures in copper, Von Holle & Trimble (1976) determined temperatures by using the ratio of detector signals obtained at different wavelengths. They believed this would reduce the likelihood of error due to changes in emissivity and the effects of surface processes, and yield a relatively unbiased estimate of the post-shock temperature. Unfortunately, if the signal/noise ratio is low, as at low temperatures and especially for metals where the emissivity is low, small errors in the measurement of detector output can lead to large changes in the ratio of the signals from the two detectors and totally unreasonable temperatures. In these experiments the ratio technique did not prove useful, although for the higher pressure shots the calculated ratio temperatures were, in fact, in good agreement with the separate brightness temperatures.

The effects of non-uniform heating may be more severe; they were estimated by calculating the temperature that would be measured if 10 per cent of the surface were 100 or 200 K hotter than the remaining 90 per cent. For a 100° excess, the discrepancy between 'measured' and mean temperatures is close to the accuracy of measurement, i.e. ± 10°. However, for a 200° excess the differences may reach ~30° (with the measured temperature being an over-estimate), although such a temperature distribution seems rather...
unlikely because of the energy partition that would be required. Non-uniform shock heating has been clearly demonstrated in silicates where such features as adiabatic shear zones (e.g. Grady 1977), deformational twin lamellae and localized production of glass (e.g. in bronzite, Gibbons 1974) have been observed. This non-uniform heating is believed to contribute to the complete loss of strength once the Hugoniot elastic limit is exceeded. The effect of surface processes such as jetting can also bias the temperature measurements; these will presumably be more important in metals where the optical depth is of the order of angstroms, than in silicates where it is microns. To reduce the likelihood of jetting, the sample surfaces were polished, but not to a high gloss as this would reduce the emissivity. Surface processes are probably more important in metals where the infrared optical depths are ~ $10^{-10}$m, than silicates where they are microns.

**Observations**

(a) **Stainless Steel-304 and Aluminium-2024**

Typical detector output records are shown in Fig. 7: these are for the InSb detector, but records obtained with the HgCdTe detector were essentially similar except for the longer rise time and the fact that the amplifier saturated prior to the peak $T_2$. The origin of this later peak is unclear: the temperature correlates well with pressure, shock and free surface velocities for both stainless steel and aluminium. Since the correlation curves for the two materials are distinct even if a common temperature calibration curve is used it appears to be a material property, rather than some effect common to all shots, such as the compression of residual gas within the sample chamber (it was not observed in shots on silicates at similar pressures) and may be related to localized heating on break up of the sample or to the action of radial release waves.

The temperatures determined are listed in Tables 4 and 5: agreement between the two wavelength ranges is good, with the difference not exceeding the estimated uncertainty in

![Figure 7. InSb detector output records for aluminium and stainless steel. $T_1$ = residual temperature, $T_2$ = peak temperature, $A$ = air shock.](http://gji.oxfordjournals.org/)
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Table 4. Measured post-shock temperatures in stainless steel-304.

<table>
<thead>
<tr>
<th>Pressure, GPa</th>
<th>InSb</th>
<th>HgCdTe</th>
<th>Peak (InSb)</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>SS</td>
<td>BB</td>
<td>SS</td>
</tr>
<tr>
<td>11.5</td>
<td></td>
<td>80</td>
<td>60</td>
</tr>
<tr>
<td>11.7</td>
<td>110</td>
<td>75</td>
<td>50</td>
</tr>
<tr>
<td>13.0</td>
<td>125</td>
<td>80</td>
<td>60</td>
</tr>
<tr>
<td>14.5</td>
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<td>130</td>
<td>85</td>
</tr>
<tr>
<td>16.0</td>
<td>145</td>
<td>100</td>
<td>95</td>
</tr>
<tr>
<td>23.0</td>
<td>195</td>
<td>130</td>
<td>1530</td>
</tr>
<tr>
<td>24.2</td>
<td></td>
<td>200</td>
<td>130</td>
</tr>
<tr>
<td>43.0</td>
<td>325</td>
<td>230</td>
<td></td>
</tr>
<tr>
<td>50.0</td>
<td>355</td>
<td>250</td>
<td>1820</td>
</tr>
</tbody>
</table>

SS = calibration using stainless steel.
BB = blackbody temperature.

Uncertainties in temperature: ± 15° below 150°, ± 10° above 150°. \( T_0 = 24°C \).

Table 5. Measured post-shock temperatures in aluminium-2024.

<table>
<thead>
<tr>
<th>Pressure, GPa</th>
<th>InSb</th>
<th>HgCdTe</th>
<th>Peak (InSb)</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>AL</td>
<td>BB</td>
<td>AL</td>
</tr>
<tr>
<td>10.5</td>
<td>125</td>
<td>50</td>
<td>140</td>
</tr>
<tr>
<td>11.5</td>
<td>135</td>
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<td>15.0</td>
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<td>25.0</td>
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<tr>
<td>27.0</td>
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</tr>
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<td>32.5</td>
<td></td>
<td>260</td>
<td>127</td>
</tr>
<tr>
<td>33.0</td>
<td></td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

AL = calibrated with aluminium.
BB = equivalent blackbody temperature.

Estimated uncertainties: ± 20°C below 200°C, ± 10°C above 200°C. \( T_0 = 24°C \).

Measurement. Also tabulated are the corresponding blackbody temperatures, which should place a lower bound on the residual temperatures; the values for aluminium are probably too low, since the maximum reported emissivity for roughened oxidized aluminium-2024 is ~ 0.4 (Touloukian & Ho 1972) which would imply temperatures of 90 to 260°C.

(b) SILICATES

Typical records obtained using the InSb and HgCdTe detectors are shown in Figs 8 and 9. Unlike quartz and bronzite, forsterite has very different records for the two wavelength ranges, probably because it transmits up to 20 per cent in the InSb range; the InSb temperatures may thus be contaminated by radiation from the sample-driver interface. The records
for quartz have a characteristic short duration 'flash' which is probably associated with triboluminescence, a phenomenon documented in quartz shocked to similar pressures by Nielson et al. (1961); the blackbody temperatures corresponding to this peak are unlikely to be significant. Blackbody and corrected temperatures for the silicates studied are listed in Tables 6 to 8.

Comparison of observed and calculated temperatures

(a) Calculational Techniques

The most widely used method of calculating shock, and hence post-shock, temperatures is probably that developed for metals by Walsh & Christian (1955) who derived a differential equation relating the shock temperature \( T_H \) to the volume on the Hugoniot \( u_H \) and material properties. This equation has the general solution

\[
T_H(u_H) = T_0 \exp \left( - \int_{u_0}^{u_H} b(v) \, dv \right) + \exp \left( - \int_{u_0}^{u_H} b(v) \, dv \right) \int_{u_0}^{u_H} f(v) \exp \left( \int_{v_0}^{v} b(u) \, du \right) \, dv. \tag{2}
\]

Here \( T_0 \) and \( u_0 \) are the initial temperature and volume, \( C_v \) the specific heat at constant volume,

\[
b = \frac{1}{C_v} \left( \frac{\partial P}{\partial T} \right)_v = \gamma
\]
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Figure 9. HgCdTe detector output records for the silicates studied. $T$ = residual temperature.

Table 6. Measured post-shock temperatures in quartz.

<table>
<thead>
<tr>
<th>Pressure, GPa</th>
<th>InSb Flash</th>
<th>BB</th>
<th>Corr.</th>
<th>HgCdTe Flash</th>
<th>BB</th>
<th>Corr.</th>
</tr>
</thead>
<tbody>
<tr>
<td>5.0</td>
<td>235</td>
<td>80</td>
<td>87</td>
<td>110</td>
<td>75</td>
<td>85</td>
</tr>
<tr>
<td>5.5</td>
<td></td>
<td></td>
<td></td>
<td>320</td>
<td>115</td>
<td>127</td>
</tr>
<tr>
<td>8.0</td>
<td>180</td>
<td>100</td>
<td>105</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>9.5</td>
<td>225</td>
<td>120</td>
<td>125</td>
<td>340</td>
<td>177</td>
<td>187</td>
</tr>
<tr>
<td>10.8</td>
<td>245</td>
<td>155</td>
<td>162</td>
<td>377</td>
<td>185</td>
<td>195</td>
</tr>
<tr>
<td>11.5</td>
<td>252</td>
<td>160</td>
<td>170</td>
<td>390</td>
<td>242</td>
<td>255</td>
</tr>
<tr>
<td>15.0</td>
<td>340</td>
<td>177</td>
<td>187</td>
<td>425</td>
<td>250</td>
<td>265</td>
</tr>
<tr>
<td>15.5*</td>
<td></td>
<td></td>
<td></td>
<td>–</td>
<td>160</td>
<td>180</td>
</tr>
<tr>
<td>17.5</td>
<td>377</td>
<td>185</td>
<td>195</td>
<td>706</td>
<td>320</td>
<td>340</td>
</tr>
<tr>
<td>19.5</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>20.0</td>
<td>390</td>
<td>242</td>
<td>255</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>21.5</td>
<td>425</td>
<td>250</td>
<td>265</td>
<td></td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

* This was a very faint record, and may not be reliable. Uncertainties: ±10°C below 100°C, ±5°C above 100°C. $T_0 = 24^\circ$C.
Table 7. Measured post-shock temperatures in forsterite.

<table>
<thead>
<tr>
<th>Pressure, GPa</th>
<th>InSb</th>
<th>HgCdTe</th>
</tr>
</thead>
<tbody>
<tr>
<td>7.5</td>
<td>180</td>
<td>&lt;50°</td>
</tr>
<tr>
<td>9.6*</td>
<td>237</td>
<td>105</td>
</tr>
<tr>
<td>15.0†</td>
<td>260</td>
<td>136</td>
</tr>
<tr>
<td>18.0</td>
<td></td>
<td></td>
</tr>
<tr>
<td>20.2</td>
<td>285</td>
<td>140</td>
</tr>
<tr>
<td>21.0</td>
<td></td>
<td></td>
</tr>
<tr>
<td>24.0</td>
<td></td>
<td></td>
</tr>
<tr>
<td>24.5</td>
<td>270</td>
<td>148</td>
</tr>
<tr>
<td>28.0</td>
<td>300</td>
<td>156</td>
</tr>
</tbody>
</table>

* Two peaks (175, 237°C); residual temperature corresponds to the difference between the levels after the second and first peaks.
† Two peaks (135, 260°C); residual temperature estimated as before.

Estimated uncertainties: ± 10°C below 100°C; ± 5°C above 100°C. \( T_0 = 24°C \).

Table 8. Measured post-shock temperatures in Bamble bronzite.

<table>
<thead>
<tr>
<th>Pressure, GPa</th>
<th>InSb</th>
<th>HgCdTe</th>
</tr>
</thead>
<tbody>
<tr>
<td>6.0</td>
<td></td>
<td>≤50°</td>
</tr>
<tr>
<td>10.3</td>
<td>123</td>
<td>100</td>
</tr>
<tr>
<td>11.0</td>
<td></td>
<td></td>
</tr>
<tr>
<td>14.8</td>
<td></td>
<td></td>
</tr>
<tr>
<td>15.5</td>
<td>185</td>
<td>147</td>
</tr>
<tr>
<td>20.7</td>
<td>225</td>
<td>200</td>
</tr>
<tr>
<td>21.5</td>
<td></td>
<td></td>
</tr>
<tr>
<td>25.0</td>
<td>200</td>
<td>175</td>
</tr>
<tr>
<td>26.0</td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

Uncertainties: ± 10°C below 100°C, ± 5°C above 100°C. \( T_0 = 24°C \).

where \( \gamma = \text{Grüneisen's parameter and} \)

\[
f(u) = \frac{dp}{du} \left( \frac{v_0 - v}{2} \right) = \frac{P}{2}
\]
evaluated at pressure, volume points \((P, v)\) along the Hugoniot.

A common assumption is that \( \gamma \) has the form

\[
\gamma = \gamma_0 \left( \frac{v}{v_0} \right)^n
\]

where \( n \geq 1 \); if \( n = 1 \), then \( b \) is constant and equation (2) reduces to the form actually derived by Walsh & Christian. Equation (2) is reduced to a difference equation and solved iteratively along the Hugoniot for \( T_H \).
The derivation of equation (2) is based on two main assumptions: first, that an ordinary fluid-type equation of state is valid, which ignores the effects of rigidity or elasto-plastic work and, secondly, that thermodynamic equilibrium exists in states behind the shock front. In addition, it is strictly valid only where the shock state is reached by a single step and not in the two-wave region associated with the Hugoniot elastic limit or phase changes; it must thus be modified for application to silicates such as those studied here. In the case of quartz, Wackerle (1962) modified the approach of Walsh & Christian by calculating temperatures along an 'equilibrium' Hugoniot obtained from a segmented linear fit to the plot of effective shock velocity $U^*$ against effective particle velocity $u^*$, where $U^*$ and $u^*$ are given by

$$U^* = v_0 [P_H/(v_0 - v_H)]^{1/2}$$
$$u^* = [P_H(v_0 - v_H)]^{1/2}$$

where $v_H$ is the volume corresponding to a shock pressure $P_H$. (These represent the true velocities only at high enough pressures where the two-wave structure is overdriven.) Such equilibrium Hugoniots may be defined for quartz and forsterite using Wackerle's (1962) data for quartz and the single crystal datum reported in Appendix B, the data for polycrystalline forsterite (McQueen 1968) and 4 per cent porous polycrystalline forsterite (Ahrens et al. 1971). However, the data for bronzite are so scattered, probably because of varying porosity, that no satisfactory equilibrium Hugoniot can be derived; instead, Walsh & Christian temperatures were calculated along Hugoniots derived from Birch-Murnaghan adiabats using the constants of Table 9 and an initial specific volume of 0.304 cm$^3$g$^{-1}$ (for details of this method see, e.g. Davies (1974)). A theoretical Hugoniot was also used in calculations involving forsterite.

Table 9. Some constants relevant to the calculation of temperatures in shocked silicates.

<table>
<thead>
<tr>
<th></th>
<th>$\rho_0^1$</th>
<th>$\gamma_0$</th>
<th>$\theta_D^1$</th>
<th>$\bar{m}$</th>
<th>$K_{os}$</th>
<th>$K'_{os}$</th>
</tr>
</thead>
<tbody>
<tr>
<td>Quartz</td>
<td>0.377</td>
<td>0.703$^2$</td>
<td>1050</td>
<td>20.03</td>
<td>37.7$^2$</td>
<td>6.4$^2$</td>
</tr>
<tr>
<td>Forsterite</td>
<td>0.310</td>
<td>1.17$^{11,3}$</td>
<td>900</td>
<td>20.12</td>
<td>126.7$^3$</td>
<td>5.37$^3$</td>
</tr>
<tr>
<td>Bronzite</td>
<td>0.298$^4$</td>
<td>0.907</td>
<td>950</td>
<td>20.96</td>
<td>103.5$^5$</td>
<td>5.3$^4$</td>
</tr>
<tr>
<td></td>
<td>0.307</td>
<td>1.56$^6$</td>
<td></td>
<td></td>
<td>105.0$^7$</td>
<td></td>
</tr>
</tbody>
</table>

$\gamma_0$ = thermodynamic Grüneisen parameter.
$\theta_D$ = Debye temperature.
$\bar{m}$ = mean atomic weight.
$K_{os}$ = zero pressure adiabatic bulk modulus.
$K'_{os} = \left(\frac{3K_{os}}{\pi^2} \right) T$.

1 Debye temperatures derived from fitting specific heat data from J.A.N.A.F. Tables (1971).
2 Values from Anderson et al. (1968).
5 Frisillo & Barsch (1972) [Mg$_{0.8}$Fe$_{0.2}$SiO$_4$].
6 Chung (1971); preferred value yielding better fit to Hugoniot data.

Note: $\gamma/u$ was generally assumed constant.
An alternative approach was described in detail by Ahrens et al. (1969b). The increase in internal energy $\Delta E_H$ of a material shocked to a state with volume $v_H$ and pressure $P_H$ is given by the Rankine–Hugoniot conservation equation and is equated to the increase in internal energy resulting from isothermal compression at $T_0$ from an initial volume to a final volume $v_H$ plus isovolumic heating to the shock temperature $T_H$. In this case

$$\Delta E_H = \frac{1}{2}(P_H + P_0)(v_0 - v_H) = \int_{v_0}^{v_H} \left( \frac{T \gamma C_v}{v} - P \right) dv + \frac{P_H - P_T}{b}$$

(5)

where $P_T$ is the pressure on the isotherm, at volume, $V_H$.

This is solved for $P_T$, and $T_H$ determined from the expression for the pressure difference between the Hugoniot and the isotherm.

$$P_H = P_T = b \int_{T_0}^{T_H} C_v dT$$

(6)

where $b = \gamma / \nu$ is assumed constant.

This formulation is easily adapted for inclusion of the Hugoniot elastic limit, and for calculating shock temperatures in the high-pressure regime for materials which undergo phase changes. Both quartz and bronzite do so in the pressure range investigated (starting at ~15 and ~13.5 GPa respectively) but neither is complete until $\geq 40$ GPa. The highest pressure measurements here are thus in the mixed phase regime – in quartz at 20 GPa the phase change is $\leq 25$ per cent complete (Grady, Murri & Fowles 1974). Calculations in the mixed phase regime are extremely complicated, the Hugoniot state may be assumed to be a mixture of high- and low-pressure phases in thermal and mechanical equilibrium and equations derived which can be solved numerically for the mass fraction of transformed material and the Hugoniot temperature at a series of points on the mixed phase Hugoniot (Ahrens et al. 1969b). However, in view of the uncertainties in the mechanism of the phase change, and in bronzite its nature (the high-pressure phase is probably majorite (Smith & Mason 1970), but may be ringwoodite plus stishovite (Ahrens & Gaffney 1971)) and the pressure range studied, the effects of the phase change were ignored, except in so far as they changed the shape of the equilibrium Hugoniot for quartz. The calculation of post-shock temperatures in the mixed phase regime is further complicated by the behaviour of the high-pressure phase on release, when it may transform to an amorphous form (Grady et al. 1974).

The calculational methods derived by Walsh & Christian and Ahrens et al. both yield values for Hugoniot temperatures, and the residual temperatures must be determined from these. It is generally assumed that the release is isentropic, in which case the residual temperature $T_R$ is given by

$$T_R = T_H \exp \left( \int_{V_R}^{V_H} b(v) dv \right)$$

(7)

where $V_R =$ volume on release, often assumed equal to $V_0$. For cases where the release path is known, the residual temperature may be calculated directly, as described by Gibbons (1974): the energy in the Hugoniot state is equated to the change in internal energy due to the rise in temperature from the initial value $T_0$ to the residual value $T_R$ plus the energy change on release. Thus

$$\frac{1}{2}(P_H + P_0)(V_0 - V_H) = \int_{V_H}^{V_R} [P dv]_{\text{release}} + \int_{T_0}^{T_R} C_p dT$$

(8)

where $V_R$ is the volume on release and $C_p$ the specific heat at constant pressure.
Post-shock temperatures in minerals

Figure 10. Comparison of observed and calculated temperatures in (a) stainless steel and (b) aluminium. The dashed curves are the Hugoniot temperatures calculated using the Walsh & Christian technique, and the solid curves the corresponding release temperatures. The dotted curve in (a) represents residual temperatures calculated directly assuming elasto-plastic release. The dot-dash and dotted curves in (b) are respectively the shock and post-shock temperatures obtained using the treatment of Foltz & Grace (1969).

(b) RESULTS FOR ALUMINIUM AND STAINLESS STEEL

Since the Walsh & Christian approach was developed specifically for application to metals (in the absence of phase changes) where the Hugoniot elastic limits are low (≤ 1.0 GPa), the application of this technique should yield results in good agreement with the experimental observations. Fig. 10(a), (b) show the values of shock (dashed lines) and post-shock temperature (solid curves) calculated for stainless steel-304 and aluminium-2024 using this approach; they are the same as those given by McQueen et al. (1970). Also plotted are the observed values and, as can be seen, there is practically no agreement. In fact, the measured residual temperatures are, at low pressures, in excess of the calculated Hugoniot temperatures. Discrepancies between observed and theoretical residual temperatures in metals have also been reported by other workers (e.g. Von Holle & Trimble 1976) using similar experimental techniques, and the question of the validity of the measurement arises. Certainly, the measured values may be too high because of surface processes or changes in emissivity — although they are consistent in two wavelength ranges — but the fact that they tend towards the theoretical values with increasing pressure suggests that the theory does not include some effect that dominates at low pressures. (Allowing for reasonable emissivity changes, the observed values are, in fact, still higher than those calculated.) The observed temperatures for steel may at first seem high when compared to the small amount of heating apparently observed in steel containers used in recovery experiments; however, these containers are not examined immediately, and the initial post-shock temperatures will

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quickly decay owing to thermal conduction. This is borne out by the observations of Schneider & Stilp (1977) who used thermocouples to measure the temperatures within large steel targets as a function of time and distance from the centre of impact by small steel cylinders and lead bullets. The time resolution of their measurements was only 50 ms, and they found that the temperature decayed rapidly as a function both of time and of distance away from the impact. Since in the current experiments the temperature at the centre of impact is observed within 1 μs of the shock wave arriving at the free surface, the high observed temperatures are not inconsistent with the maximum increase of 18°C observed 1.2 (or approximately the crater radius) cm from the impact centre by Schneider & Stilp.

One obvious omission from the Walsh & Christian formalism is the effect of elasto-plastic work. Although the metals have low Hugoniot elastic limits, they retain some rigidity after yielding, and may undergo stress-hardening. The latter was reported by Fowles (1961), who demonstrated that an elasto-plastic equation of state should be used for aluminium shocked to pressures up to ~15 GPa. This concept was developed in detail by Lee & Liu (1967) and Lee & Wierzbicki (1967) who developed a treatment allowing direct calculation of temperature along the Hugoniot curve which incorporates the effects of material strength and finite anisotropic strain. Foltz & Grace (1969) carried out the analysis for polycrystalline aluminium having a yield strength comparable to that for aluminium-2024; whilst their analysis may not be strictly valid for the alloy used in these experiments, a comparison of their values for Hugoniot temperature (the dash-dot line in Fig. 10(b)) and the measured residual temperatures in interesting. As can be seen, their calculated values are considerably in excess of the Walsh & Christian values at low pressures, but converge with them at higher pressures, which is precisely the behaviour observed in the measured residual temperatures. Although the release path is not certain, it has been proposed that for metals the release from shock pressure \( P_H \) occurs in two stages: first, an elastic release (isentropic) to a pressure \( P_H - 2Y \), where \( Y \) is the Hugoniot elastic limit, followed by plastic release parallel to the Hugoniot (see, e.g. Fowles 1961; Al'tschuler 1965). The post-shock temperatures indicated by the dotted curve in Fig. 10(b) were derived from Foltz & Grace's Hugoniot temperatures assuming this form of release path with \( Y = 0.8 \) GPa. These values are in reasonable agreement with experimental observations (which may be slightly high due to changes in emissivity). Unfortunately, a similar calculation cannot be made for an alloy like stainless steel where the constants required are not really defined. However, the relationship between the observed and Walsh & Christian values (Fig. 10(a)) is similar to that observed for aluminium, and plastic deformation of the lattice has been observed in recovery experiments for pressures up to 50 GPa (e.g. Murr 1975; Smith 1958). The post-shock temperatures indicated by the solid line in Fig. 10(a) were calculated from the Hugoniot temperatures assuming a simple isentropic release path; some allowance for elasto-plastic behaviour may be made by assuming a release path as indicated above and calculating the residual temperatures directly using 8. This yields the dotted line in Fig. 10(a), but the agreement between experiment and theory is not substantially improved.

It is concluded that elasto-plastic work, which is not included in the Walsh & Christian formalism, causes significant heating at low pressures resulting in large differences between measured and calculated temperatures. However, at higher pressures (<30 GPa for aluminium, <50 GPa for stainless steel) the Walsh & Christian approach appears to predict values close to those measured experimentally.
temperatures calculated directly from release adiabat data by Mashimo et al. (1979) is very good, and this method of calculation is certainly to be preferred. However, the values calculated by Wackerle (1962) corrected for an initial temperature of 24°C are also remarkably close to those observed except for the lowest and highest pressure points. The agreement may be improved by determining residual temperatures from the Hugoniot temperatures using minimum release volumes estimated from the free surface and particle velocities given by Wackerle following the method described by Lyzenga & Ahrens (1978) or the release volumes given by Grady et al. (1974).

Temperatures calculated using the method of Ahrens et al. (1969b) using values of 6 and 8 GPa for the Hugoniot elastic limit (the lower and upper bounds given by Wackerle for z-cut quartz) were much too low; the observed values are in fact greater than the calculated shock temperatures. Similar discrepancies were also noticeable for bronzite and forsterite, and may arise in the determination of the isotherm; it suggests this calculational technique is inadequate in the pressure range studied.

For bronzite, none of the existing calculational techniques produced adequate agreement between theoretical and observed post-shock temperatures (Fig. 12) and since the observed values are generally in excess of the calculated shock temperatures some process must occur which results in greater heating on compression. (This may be related to the effects of sample porosity: the apparent temperature drop between 20.7 and 25.0 GPa for InSb measurements probably reflects the fact that the latter sample had a lower density of fine cracks than
Figure 12. Observed and calculated temperatures for Bamble bronzite. Solid curves are post-shock temperatures, broken curves Hugoniot temperatures. (a) A, A'': theoretical Hugoniot $\gamma = 0.907$, B, B': theoretical Hugoniot, $\gamma = 2.5$ on compression, 1 on release, C, C': actual Hugoniot, $\gamma = 0.907$, Hugoniot elastic limit of 6.7 GPa (Ahrens & Gaffney 1971). (b) D, D': theoretical Hugoniot, $\gamma = 1.57$, E, E': actual Hugoniot, $\gamma = 1.57$, Hugoniot elastic limit of 6.7 GPa.

usual.) Attempts were made to improve the Walsh & Christian calculations by using a temperature, as well as volume, dependent Debye temperature in calculating the specific $C_v$ via the Debye model. However, the post-shock temperatures are relatively insensitive to these refinements and no marked improvement in agreement was produced. The calculations of residual temperature are also fairly insensitive to the behaviour of Gruneisen's parameter provided it is the same on compression and on release. A model in which the Gruneisen parameter is higher on compression than release yields higher residual temperatures. Although we cannot defend such a model in detail, this idea does in fact have a good physical basis, since the shock and release processes are certainly different. In particular, the observations of release paths lying beneath the Hugoniot (e.g. Fig. 1) and the apparent hysteresis in the shock and release process are not fully understood; they may be related to the behaviour of $\gamma$. A model in which $\gamma_0(=\gamma(V_0))$ was assumed to be 2.5 on compression and 1 on release produced somewhat better agreement, especially at high pressure, but a Gruneisen parameter of at least 3 during shock compression would be required to produce agreement at pressures less than 15 GPa. In this case the calculated release temperatures (assuming $\gamma_0 = 1$) would be too high at pressures of ~25 GPa, so a more complicated behaviour of $\gamma$ would have to be postulated. This does not seem justified since the theoretical Hugoniot does not adequately describe the effects of varying samples porosity, and the high temperatures may be largely due to this; furthermore, the effect of the phase change has been neglected — although this would provide some justification for changing the behaviour of the effective $\gamma$ at 15 GPa.

The results of the temperature calculations for forsterite are shown in Fig. 13; at this stage it should be noted that the 10 and 15 GPa measurements using the InSb detector are
Figure 13. Observed and theoretical temperatures for forsterite. (a) A, A': polycrystalline forsterite, equilibrium Hugoniot, B, B': ~4 per cent porous forsterite, equilibrium Hugoniot, C, C': theoretical Hugoniot. (b) D, D': theoretical Hugoniot, \( \tau = 2.5 \) on compression, \( \gamma = 1 \) on release, E, E': actual Hugoniot, assumed Hugoniot elastic limit 5 GPa, F: actual Hugoniot elastic limit 9 GPa shock temperatures. Note: the measured H.E.L. was 8.7 GPa.

The observed temperatures are considerably in excess of the values calculated for the non-porous polycrystalline forsterite using the equilibrium Hugoniot (curves A, A', Fig. 13(a)), and are also greater than the values calculated for the theoretical Hugoniot (C, C') although there is some indication that the measurements tend towards the latter at high pressures. The values calculated for the 'porous' equilibrium Hugoniot (B, B') are much higher than the observations except for the doubtful InSb points; this is not surprising since the samples were only \( \approx 1 \) per cent porous, and not 4 per cent. As with bronzite, a better fit to the observations may be obtained using the Walsh & Christian approach but assuming that \( \gamma_0 \) is higher on compression than release; a fairly good fit resulted from using the theoretical Hugoniot with \( \gamma_0 = 2.5 \) on compression and \( \gamma_0 = 1 \) on release. However, although temperatures calculated by the conventional methods of Walsh & Christian and Ahrens et al. are much lower than the observations at low pressure, they converge at higher pressures, which is reminiscent of the behaviour observed in stainless steel and aluminium and ascribed to elasto-plastic effects. A detailed investigation of the process of dynamic yielding in forsterite would indicate whether such effects were possible here; if the behaviour of forsterite is indeed similar to that of metals, then the release path may be approximated by the two stage elasto-plastic path that was described earlier and is sketched in Fig. 14. In this case an estimate of the release temperature may be obtained by calculating the energy difference between the shock and release paths (the shaded area in Fig. 14) and equating it to the increase in internal energy as in equation (8). This results in the temperatures shown by the heavy line in Fig. 13(b); the agreement between these values and the observations is good at \( \approx 10 \) GPa, but large differences exist at higher pressures, as would be expected if the
material underwent a gradual loss of strength that was complete by ~ 25 GPa. (Experimental evidence indicates that the loss of strength is in fact complete by ~ 20 GPa; see Appendix B.) This calculation ignored the fraction of energy going into permanent deformation of the lattice, which may be estimated as follows: the free energy of a crystal is increased by about $\mu b^3/2$ per atomic length of dislocation, where $\mu$ is the shear modulus and $b$ is the average length of the Burgers vectors of the crystals. With $b \sim 5 \text{ Å}$, $\mu = 80 \text{ GPa}$, and a dislocation density of $10^{11} \text{ cm}^{-2}$, which is probably rather high for these pressures (R. Jeanloz 1978, private communication), allowing for the deformational energy results in a 15 per cent or smaller reduction in the increase in thermal energy due to shock compression and release. This is equivalent to a 6°C (or smaller) decrease in release temperatures from the values plotted in Fig. 13(b).

**Discussion**

The experimental technique described here has proved useful for measuring post-shock (brightness) temperatures in a variety of metals and silicates. Initial experiments produced results which are, in general, consistent at two wavelength ranges (from 4.5 to 5.75 μm and from 7 to 14 μm) and for different samples of the same material shocked to similar pressures. Although the consistency of the results suggests that the measured temperatures are indeed representative of the residual temperatures in the shocked samples, uncertainties still exist in the precision of the results for the metals because of the lack of information on the effect of surface process, which are likely to be important in metals, and changes in emissivity under shock conditions. In particular, few data are available on the emissivity of the silicates studied, even under normal conditions; and if such data were available more accurate correction factors could be estimated, although the possibilities of non-equilibrium radiation with an emissivity greater than unity and changes in emissivity under shock conditions would remain. Experimental investigation of the latter, if possible, would be most valuable. One effect that may influence the measurements of residual temperatures in
aluminium is the ejection of material from the surface; this has been studied by Asay, Mix & Perry (1976) for pressures of ~25 GPa and is quite significant. A further study by Asay (1977) showed that material ejection was highly dependent on the rise-time of the shock wave, so the effect of material ejection on temperature measurement might be investigated by determining residual temperatures for different shock rise-times.

Measured temperatures in quartz, which is known to undergo a complete loss of strength after the Hugoniot elastic limit is exceeded, were in good agreement with those calculated by Mashimo et al. (1979) from release adiabat data, and in reasonable agreement with those derived from the equilibrium Hugoniot of Wackerle (1962) using estimated release volumes. These calculations did not include the affects of the quartz to stishovite phase change which begins at ~15 GPa; however there was only a slight change in the rate of increase of measured temperature with pressure in this region.

For both the metals studied, and for forsterite and bronzite, the measured temperatures were considerably in excess of those predicted using conventional fluid rheology assumptions, although the absence of release adiabat data for forsterite and bronzite meant that a comparison with temperatures calculated in the optimal manner (i.e. using equation (8)) was not possible. In all these cases some process is required which leads to higher calculated shock temperatures, especially at low pressures, and less energy loss on release. For metals, this is probably elastoplastic work, which dominates at low pressures: reasonable agreement was found between the elasto-plastic calculations of Foltz & Grace (1969) and measurements for aluminium. An attempt was made to model such a process in bronzite and forsterite by allowing Grüneisen’s parameter to be higher on compression than on release, and for forsterite reasonable agreement was obtained with $\gamma_0 = 2.5$ on compression, $\gamma_0 = 1$ on release (note that under standard conditions the thermodynamic Grüneisen’s parameter is 1.17). No strict physical justification for postulating a different $\gamma$ for shock compression from that on unloading is attempted here; however, this type of behaviour does provide better agreement between theory and observations although the model is somewhat non-unique. The differences between observed and predicted release adiabats and the hysteresis observed in shock unloading suggest that some change in material properties may occur prior to release from the shock state, and thus irreversible behaviour of the anharmonic properties is not incredible. Another possibility is that elastoplastic work may also be important: although quartz has been observed to undergo a complete loss of material strength beyond the Hugoniot elastic limit, the same has not been demonstrated for forsterite. Calculations of residual temperatures in which it was assumed that forsterite behaved in an elasto-plastic manner similar to that proposed for metals yielded values that agreed well with observations at 10 GPa but were much higher at higher pressures. This suggests that forsterite undergoes a much more gradual loss of material strength beyond its elastic limit than does quartz, and that elastoplastic effects may be important at pressures below ~25 GPa.

Classification of shock metamorphism such as that of Stöffler (1971) are based on a detailed analysis of the physical characteristics of the specimen (such as the amount of fracturing, undulatory extinction or presence of glass.) Estimates of the temperature range corresponding to each class of metamorphism have then been made using calculated values of the shock and post-shock temperatures: Stöffler used the values of Wackerle (1962) for quartz and Ahrens et al. (1969b) for feldspars. The results of the present study indicate that the former are fairly reliable below ~20 GPa (when corrected for the right initial temperature), but the calculation technique of Ahrens et al. yielded temperatures that were substantially lower than those observed, so that the values used by Stöffler for feldspars are also likely to be low. Indeed, observations of recovered shocked samples suggest
that this is the case both in feldspars (e.g. Gibbons 1974) and dunite (e.g. Reimold & Stöffler 1978). Measurements of the electrical conductivity of shocked dunites (Schulein, Hornemann & Stöffler 1978) show increases that are also more easily explained if the actual temperatures are higher than those calculated. The main implication of this study for impact metamorphism is that temperatures in forsterite and bronzite appear considerably higher than the theoretical values, especially for porous samples, and so the effects of impact metamorphism on basic rocks such as lunar basalt may differ from those expected on the basis of theory and lead to erroneous conclusions regarding the nature of the impact. Since the partitioning of energy during impact processes is important in calculating the thermal history of an accreting planet, residual temperature data such as those presented here could provide useful constraints on the temperature distribution during accretion.

One important assumption used in the temperature calculations was that equilibrium existed behind the shock-front. Temperatures measured in these experiments were generally higher than those calculated, but other factors leading to errors in the calculation make it hard to assess the validity of this assumption. A systematic investigation of differences between shock effects seen in shock recovery experiments and in naturally shocked samples where the peak pressures are of longer duration could help clarify this point. Such differences certainly exist — for example, the substantial amount of coesite observed in natural, but not shock-recovery, specimens (Stöffler 1972) and might yield important information on the nature of equilibrium in shock processes.

It has been emphasized that the release path is of critical importance to the determination of post-shock temperatures, yet it has generally been assumed that the release is isentropic; if this is not the case, then discrepancies between measurement and theory are not surprising. Recently Kieffer & Delany (1979) have placed new and useful constraints on the assumptions of isentropic flow in solids under decompression. Viscous heating on release would lead to higher residual temperatures.

Release adiabats which lie below the Hugoniot (see Fig. 1) lead to higher residual temperatures in better agreement with the observations, yet it seems strange that if the material is essentially unchanged by shock compression and release, it should occupy a

Figure 15. Two release paths having the same initial slope and area underneath yet different final volumes.
Post-shock temperatures in minerals

Table 10. Estimated release volumes.

<table>
<thead>
<tr>
<th>Material</th>
<th>u₀ (cm³g⁻¹)</th>
<th>α_p (GPa)</th>
<th>Pressure, GPa</th>
<th>Volume, cm³g⁻¹</th>
</tr>
</thead>
<tbody>
<tr>
<td>Al-2024</td>
<td>0.359</td>
<td>8 × 10⁻⁵</td>
<td>5</td>
<td>0.3776</td>
</tr>
<tr>
<td>SS-304</td>
<td>0.127</td>
<td>5 × 10⁻⁵</td>
<td>10</td>
<td>0.3782</td>
</tr>
<tr>
<td>Quartz</td>
<td>0.377</td>
<td>3.5 × 10⁻⁵</td>
<td>15</td>
<td>0.3799</td>
</tr>
<tr>
<td>Forsterite</td>
<td>0.310</td>
<td>2.6 × 10⁻⁵</td>
<td>20</td>
<td>0.3107</td>
</tr>
<tr>
<td>Bronzite</td>
<td>0.302</td>
<td>2.4 × 10⁻⁵</td>
<td>25</td>
<td>0.3032</td>
</tr>
</tbody>
</table>

* Theoretical zero-pressure density (Ahrens & Gaffney 1971). Actual initial volume ~0.304 cm³g⁻¹.
† No detectable temperature change.

smaller volume on release when it is hotter than it was initially. Measurements of the parameters on release adiabats are often made at relatively high pressures (P_H > P_R > 0, where P_H is the shock pressure and P_R the point at which release measurements are made) and so yield information on the initial slope of the release adiabat which may then be extrapolated to yield the release volume. Such an extrapolation is illustrated schematically in Fig. 15 by the dashed line: this yields a release volume smaller than u₀; however, the actual path may be as shown by the dotted line. Since the area between the Hugoniot and the two release curves is the same, they will give the same residual temperatures if equation (8) is used. Assuming that the material is unchanged, then the release volume may be estimated from the post-shock temperature and the mean thermal expansion coefficient α_p, and can provide additional constraints on the release path. Table 10 lists the estimated release volumes for the materials studied: in fact, for the silicates and stainless steel they are not very different from the initial volumes.

It was hoped that this study would yield some definite information on the behaviour of Grüneisen's parameter, which is critical for the reduction of shock wave data to the form needed for comparison with density—depth profiles within the Earth. However, post-shock temperatures are not very sensitive to the behaviour of γ (unless it is different on shock compression and release) and appear more greatly influenced by the release path, in particular the release volume. There seems to be no need to postulate unusual behaviour of γ for quartz to ~ 25 GPa, although the high observed values of residual temperatures in forsterite and bronzite suggest that present rheological models used in shock temperature calculations, at least below ~ 25 GPa, are either inadequate, or the effective Grüneisen parameter is high (γ₀ ~ 2.5) on compression and decreases to near its zero pressure value (γ₀ ~ 1) upon adiabatic release.

Acknowledgments

The authors thank Raymond Jeanloz and Gregory Lyzenga for many interesting and stimulating discussions; they are also indebted to Wayne Miller and Victor Nenow, who designed and built the variable gain amplifier used with the InSb detector, and to the staff of the Santa Barbara Research Centre, Goleta, California for their advice concerning detector operation. We thank D. Stöffler for both his comments on impact metamorphism, and the
References


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Appendix A: detector parameters

Both detectors were purchased from the Santa Barbara Research Centre, Goleta, California.

(a) InSb

The InSb detector used was a circular chip 1 mm in diameter having a detectivity of $5 \times 10^{10}$ cm Hz$^{1/2}$ watt$^{-1}$ when operated at 77 K. It was operated with a fast matched pre-amplifier (SBRC Model A-230) consisting of a current mode operational amplifier with a feedback resistance of 1 kΩ and a non-inverting voltage mode post-amplifier; this stage had a gain of 500, and upper and lower 3 dB frequencies of 20 MHz and 1.35 kHz respectively. For use in measuring post-shock temperatures an additional amplifier with variable gain (from 1000 to 30000) was used (Raikes 1978). The minimum system rise time of $\sim 0.1 \mu$s is controlled by the detector chip itself. Sapphire windows were used with this detector.

(b) HgCdTe

The chip used had an area of $2 \times 10^{-2}$ cm$^2$, a detectivity of $6.94 \times 10^9$ cm Hz$^{1/2}$ watt$^{-1}$ and a rise time of $\sim 100$ ns. It was used with a matched amplifier, SBRC Model A-120, having a gain of 1000. The amplifier consisted of an AC coupled voltage mode amplifier plus a 499 Ω load resistor and circuitry to produce the bias current of 10 mA required by the detector; its upper and lower 3 db frequencies were 10 MHz and 50 Hz respectively. The rise time of the detector-amplifier system is $\sim 0.05 \mu$s; however, for operation at low signal levels it was found to produce an unacceptable level of very-high-frequency noise, and so had to be operated with a filter which raised the rise-time to $\sim 0.75 \mu$s. Barium fluoride or Irtran-2 (Kodak) windows were used with the HgCdTe detector.

Appendix B: dynamic yielding in forsterite

A detailed study of the dynamic yielding of single crystal forsterite has not been carried out. For the purpose of the present work a single measurement of the Hugoniot elastic limit along the c axis of forsterite was undertaken in order to place some bounds on the rheological behaviour at relatively low shock stress.
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Figure B1. Static and dynamic streak images produced upon shock compression of c-cut single crystal forsterite to 17 GPa. (a) Static image as seen through streak camera. (b) Streak image demonstrating two-wave shock structure recorded by inclined mirror. The angle $\gamma$, is related to the value of the Hugoniot elastic limit, see, e.g. Ahrens et al. (1968).

Table B1. Equation of state data for forsterite shocked along the (001) direction (shot 431).

<table>
<thead>
<tr>
<th></th>
<th>First shock state</th>
<th>Final shock state</th>
</tr>
</thead>
<tbody>
<tr>
<td>Sample mass</td>
<td>3.1546 ± 0.0001 g</td>
<td></td>
</tr>
<tr>
<td>Bulk density</td>
<td>3.2158 ± 0.005 gm cm$^{-3}$</td>
<td></td>
</tr>
<tr>
<td>Projectile velocity</td>
<td>1.668 ± 0.002 km/s</td>
<td></td>
</tr>
<tr>
<td>Shock velocity</td>
<td>8.717 ± 0.068 km/s</td>
<td>6.554 ± 0.067 km/s</td>
</tr>
<tr>
<td>Shock density</td>
<td>3.334 ± 0.008 gm cm$^{-3}$</td>
<td>3.5716 ± 0.0041 gm cm$^{-3}$</td>
</tr>
<tr>
<td>Shock pressure</td>
<td>8.68 ± 0.67 GPa</td>
<td>17.32 ± 0.05 GPa</td>
</tr>
<tr>
<td>Free surface velocity</td>
<td>0.619 ± 0.042 km/s</td>
<td></td>
</tr>
<tr>
<td>Particle velocity</td>
<td>0.7246 ± 0.0022 km/s</td>
<td></td>
</tr>
</tbody>
</table>

Shock loading and recording were carried out with a modification of the 40 mm gun system described in Ahrens et al. (1971) and Ahrens, O'Keefe & Gibbons (1973), although the optical recording was carried out using a Model 339B Beckman and Whitley continuous writing streak camera with a recently constructed xenon light source (described in Goto et al. 1978).
The sample, machined from the same aliquot as that utilized for the post-shock temperature measurements, was mounted on a 1.5 mm thick aluminium-2024 driver plate and impacted with a 4 mm thick aluminium-2024 flyer plate. The pertinent measurements obtained in a single experiment (Fig. B1, Table B1) provide a rather unequivocal measurement of the Hugoniot elastic limit amplitude, 8.68 ± 0.67 GPa, for a 3.7 mm thick sample. The velocity of the first shock, 8.72 km/s, is close to, but slightly higher than, the values of 8.543 km/s (Kumazawa & Anderson 1969) or 8.564 km/s (Graham & Barsch 1967) measured for the longitudinal velocity in the equation (3) direction. A similar relationship between the two velocities has been reported for quartz in various orientations (Wackerle 1962; Fowles 1967). The theoretical relation of the higher order elastic constants to the velocity of finite strength elastic waves, although reported for hexagonal symmetry (Fowles 1967) has not been worked out for orthorhombic crystals.

The arrival time of the second shock, as indicated by the marked change in slope observed in the image of the inclined mirror (Fig. B1), can be interpreted in several ways depending on the rheology assumed for the sample material which has been compressed, and subsequently released, by the first shock (Wackerle 1962). Our preferred analysis is simply to assume that the second shock velocity is given by

$$\bar{U}_2 = \frac{d + U_{fs}(t_2 - t_1)}{t_2 - t_0}$$

(B1)

where $d$ is the sample thickness, $U_{fs}$ the free surface velocity and $(t_2 - t_1)$ and $(t_2 - t_0)$ are the time interval between the arrival of the first and final shock and the travel time of the second shock to the free surface respectively (Ahrens, Gust & Royce 1968). Equation (B1) ignores the interaction of the reflection of the first shock with the oncoming final shock.
front. The parameters of the final shock state (Table B1) were obtained from an impedance match solution using the equation of state parameters given by McQueen et al. (1970) for aluminium-2024. Equation (B1) yields a final shock state which agrees well with the pressure–density trajectory calculated via the Murnaghan equation using parameters obtained from the ultrasonic data. If, on the other hand, the expression for the final shock velocity given by Ahrens et al. (1973), which assumes an elastic interaction of the reflected elastic shock with the oncoming final shock, is used a state with a greater compression is calculated (Fig. B2). Although the difference between the two solutions is not large, the calculation assuming no interaction, implying a loss of strength at pressures corresponding to the final shock state, appears to be definitely more consistent with the ultrasonic data. The relatively high Hugoniot elastic limit, followed by a (gradual) loss of shear strength, is consistent with the rheological behaviour inferred from the post-shock temperature measurements (Table 7).